

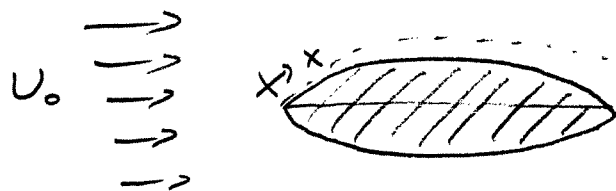
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## MECE 6373 Viscous Flow II EXAMPLE PROBLEM

empirical prediction of  $(x/L)_{lamsep}$ ,  $(x/L)_{cr}$ ,  
and  $(x/L)_{tr}$  for flow over a surface where

$$U_{\infty}(x) = U_0 [1 + 2.0(x/L) - 2.0(x/L)^2]$$

$$0 < (x/L) < 1.0$$



with  $Re_L = \frac{U_0 L}{\nu} = 1 \times 10^6$

use Thwaites method

$$\lambda(x) = \frac{9^2 \frac{dU}{dx}}{\nu} = \frac{0.45 \frac{dU}{dx}}{U^6} \int_0^x U^5(x') dx'$$

define  $x^* = x/L$

$$\lambda(x^*) = \frac{0.45 \frac{U_0}{L} [2.0 - 4.0x^*]}{U_0^6 [1 + 2.0x^* - 2.0x^{*2}]^6}$$

$$\times \int_0^{x^*} U_0^5 [1 + 2.0x^* - 2.0x^{*2}]^5 dx^*$$

$$\lambda(x^*) = \frac{0.45 [2.0 - 4.0x^*]}{[1 + 2.0x^* - 2.0x^{*2}]^6} \int_0^{x^*} [1 + 2.0x^* - 2.0x^{*2}]^5 dx^*$$

perform integral using RUNGE with

$$Y(1) = \int_0^{x^*} [1 + 2.0x^* - 2.0x^{*2}]^5 dx^*$$

$$F(1) = [1 + 2.0x^* - 2.0x^{*2}]^5$$

$x/L$  lam sep occurs when  $\lambda(x^*) = -0.09$  (ind. of  $Re_L$ !)

to find  $x/L$  cr use correlation of Wazzan

$$Re_{s^*, cr} = \text{func}(H) \quad \text{Fig (5-12)}$$

(use curve fit given in class)

note

$$\theta(x^*) = \left( \frac{\nu \lambda(x^*)}{dU/dx} \right)^{1/2} = \left[ \frac{\nu \lambda(x^*)}{U_0/L (2.0 - 4.0x^*)} \right]^{1/2}$$

and  $\delta^* = H \theta$  so

$$Re_{s^*} = \frac{U(x^*) \delta^*}{\nu} = \frac{U_0 [1 + 2x^* - 2x^{*2}]}{\nu} H(\lambda) \left[ \frac{\nu \lambda(x^*)}{U_0/L (2.0 - 4.0x^*)} \right]^{1/2}$$

$$Re_{s^*} = \sqrt{Re_L} [1 + 2x^* - 2x^{*2}] \left[ \frac{\lambda(x^*)}{(2 - 4x^*)} \right]^{1/2} H$$

where  $H = \text{func}(z) \quad E_2 (4.141)$

$$z = (0.25 - \lambda)$$

$x/L$  cr occurs when  $Re_{s^*} \geq Re_{s^*, cr}$

to find  $x/L)_{tr}$  use one-step method of Wazzan

$$Re_{x,tr} = \text{func}(H) \quad \text{Eq (5.42)}$$

where

$$Re_x = \frac{U(x)x}{\nu} = \frac{U_0 [1 + 2x^* - 2x^{*2}] L x^*}{\nu}$$

$$Re_x = Re_L x^* [1 + 2x^* - 2x^{*2}]$$

$x/L)_{tr}$  occurs when  $Re_x \geq Re_{x,tr}$

see attached program listing and results

find for  $Re_L = 1 \times 10^6$

$$x/L)_{lam\ sep} = 0.643$$

$$x/L)_{cr} = 0.421$$

$$x/L)_{tr} = 0.564$$



```
1 DO 2 I=1,N
2 Q(I)=0.0
  A=0.5
  GOTO 9
3 A=1.707107
  IF YOU NEED MORE ACCURACY, USE
C A=1.7071067811865475244
4 X=X+0.5*H
5 DO 6 I=1,N
  Y(I)=Y(I)+A*(F(I)*H-Q(I))
6 Q(I)=2.0*A*H*F(I)+(1.0-3.0*A)*Q(I)
  A=0.2928932
C IF YOU NEED MORE ACCURACY, SET
C A=0.2928932188134524756
  GOTO 9
7 DO 8 I=1,N
8 Y(I)=Y(I)+H*F(I)/6.0-Q(I)/3.0
  M=0
  K=2
  GOTO 10
9 K=1
10 RETURN
  END
```

0.000000E+00  
0.000000E+00

1

6.500000E-01

1.000000E-02

0

$x^*$	$\lambda$	H	$Re_s^*$	$Re_s^*, cr$	$Re_x$	$Re_x, tr$
.010	.008	2.567	170.	692.	1.02E+04	6.49E+06
.020	.015	2.547	235.	829.	2.08E+04	8.50E+06
.030	.021	2.530	283.	961.	3.17E+04	1.06E+07
.040	.026	2.516	321.	1086.	4.31E+04	1.27E+07
.050	.030	2.504	354.	1204.	5.47E+04	1.48E+07
.060	.033	2.494	382.	1315.	6.68E+04	1.68E+07
.070	.036	2.486	408.	1416.	7.91E+04	1.87E+07
.080	.039	2.478	432.	1508.	9.18E+04	2.04E+07
.090	.041	2.472	454.	1590.	1.05E+05	2.21E+07
.100	.043	2.467	475.	1663.	1.18E+05	2.35E+07
.110	.044	2.463	495.	1726.	1.32E+05	2.48E+07
.120	.045	2.460	514.	1780.	1.45E+05	2.59E+07
.130	.046	2.457	532.	1824.	1.59E+05	2.69E+07
.140	.047	2.455	550.	1861.	1.74E+05	2.76E+07
.150	.047	2.453	567.	1889.	1.88E+05	2.82E+07
.160	.048	2.452	584.	1909.	2.03E+05	2.86E+07
.170	.048	2.451	600.	1923.	2.18E+05	2.89E+07
.180	.048	2.450	616.	1930.	2.33E+05	2.91E+07
.190	.048	2.450	632.	1931.	2.48E+05	2.91E+07
.200	.048	2.451	648.	1926.	2.64E+05	2.90E+07
.210	.048	2.451	664.	1916.	2.80E+05	2.88E+07
.220	.048	2.452	680.	1901.	2.96E+05	2.85E+07
.230	.047	2.453	695.	1882.	3.11E+05	2.81E+07
.240	.047	2.455	711.	1859.	3.28E+05	2.76E+07
.250	.046	2.456	727.	1832.	3.44E+05	2.70E+07
.260	.046	2.458	742.	1801.	3.60E+05	2.64E+07
.270	.045	2.460	758.	1767.	3.76E+05	2.57E+07
.280	.044	2.463	774.	1730.	3.93E+05	2.49E+07
.290	.043	2.465	790.	1691.	4.09E+05	2.41E+07
.300	.042	2.468	806.	1649.	4.26E+05	2.32E+07
.310	.041	2.471	822.	1605.	4.43E+05	2.24E+07
.320	.040	2.475	838.	1559.	4.59E+05	2.14E+07
.330	.039	2.478	855.	1510.	4.76E+05	2.05E+07
.340	.038	2.482	871.	1461.	4.93E+05	1.95E+07
.350	.036	2.486	888.	1409.	5.09E+05	1.85E+07
.360	.035	2.491	905.	1357.	5.26E+05	1.75E+07
.370	.033	2.495	923.	1303.	5.42E+05	1.65E+07
.380	.031	2.500	940.	1248.	5.59E+05	1.55E+07
.390	.030	2.505	958.	1192.	5.76E+05	1.45E+07
.400	.028	2.511	976.	1136.	5.92E+05	1.35E+07
.410	.026	2.517	995.	1079.	6.08E+05	1.26E+07
.420	.023	2.523	1014.	1021.	6.25E+05	1.16E+07
.430	.021	2.530	1033.	963.	6.41E+05	1.06E+07
.440	.019	2.537	1053.	905.	6.57E+05	9.68E+06
.450	.016	2.544	1074.	846.	6.73E+05	8.76E+06
.460	.013	2.553	1095.	787.	6.89E+05	7.87E+06
.470	.010	2.562	1117.	728.	7.04E+05	7.01E+06
.480	.007	2.571	1139.	669.	7.20E+05	6.17E+06
.490	.004	2.582	1162.	609.	7.35E+05	5.37E+06
.500	.000	2.594	1187.	550.	7.50E+05	4.60E+06
.510	-.004	2.607	1212.	506.	7.65E+05	3.87E+06
.520	-.008	2.622	1239.	476.	7.80E+05	3.18E+06
.530	-.012	2.639	1267.	444.	7.94E+05	2.54E+06
.540	-.017	2.658	1298.	409.	8.08E+05	1.96E+06
.550	-.022	2.682	1330.	372.	8.22E+05	1.44E+06
.560	-.027	2.709	1366.	331.	8.36E+05	9.99E+05
.570	-.033	2.743	1407.	288.	8.49E+05	6.42E+05
.580	-.039	2.785	1452.	242.	8.63E+05	3.74E+05
.590	-.046	2.838	1504.	195.	8.75E+05	1.94E+05
.600	-.053	2.905	1566.	148.	8.88E+05	8.68E+04
.610	-.061	2.992	1640.	103.	9.00E+05	3.35E+04
.620	-.069	3.106	1732.	100.	9.12E+05	1.17E+04
.630	-.078	3.257	1848.	100.	9.24E+05	4.58E+03
.640	-.087	3.458	1996.	100.	9.35E+05	3.73E+03
.650	-.098	3.730	2192.	100.	9.46E+05	3.57E+04